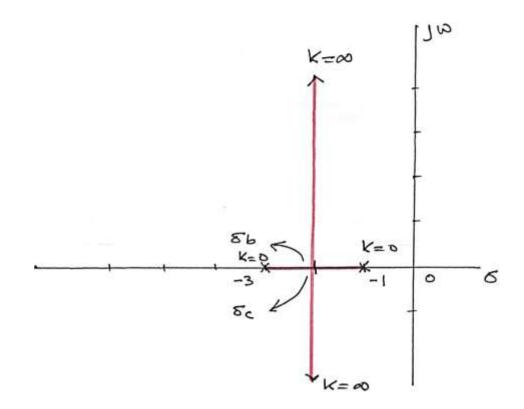
Consider a unity flb control system with
OL
$$T/F \rightarrow KQ(G)H(G) = \frac{H}{(S+1)(S+3)}$$

The rost boars is quite cary.
(i) $\beta := -1, -3, \quad 2 = 0$ is $2a+\infty$
(ii) $\beta - 2 = 2 - 0 = 2$ Ma β laci = 2
(iii) lead axis lock like $b/\omega - 1$ and -3 ordy.
(iv) There is their one break away point
Using ongle mag onton.
 $Kf(f(S))H(G) = -1$
 $\frac{K}{(S+1)(S+3)} = -1$
 $\frac{K}{(S+1)(S+3)} = -1$
 $\frac{K}{(S+1)(S+3)} = -1$
 $\frac{K}{(S+1)(S+3)} = -2S - 4 = 0$
 $S = -2$
 $Sb = -2$
(V) Centre of asymptote.
 $\delta_c = \frac{2F-2}{F-2} = \frac{-1-3-0}{2} = -2$
(vi) Angle of asymptote.
 $\delta_c = 90^{\circ}$ and 270°
(viii) There is no jw-crossing becaus $0 = 90^{\circ}, 270^{\circ}$
Thus the not beau is ont vertical time
drawn at $Sb = S = -2$

The root locus is shown below:



The root locus is a straight vertical line and therefore does not cross the jw-axis. Thus the system does not move into the unstable region (right-half plane) and is therefore always stable. Thus all the closed-loop poles lie in the left-half plane.

Now let us add a pure gain integrator in the forward path with transfer function 1/s. The openloop transfer function then takes the form: $KG(s)H(s) = \frac{K}{s(s+1)(s+2)}$. This is our Example 4. To draw the root locus we follow the rules:

Now if we add a pure integrator with T/F
of 1/s in the porward path we get a OLT/F
of the system as:
$$OLT/F \longrightarrow Kq(s)H(s) = \frac{k}{s(s+1)(s+3)}$$

To draw its not locus we have the following
requirements.
(i)
$$\beta: 0, -1, -3, 2: \text{NU} = 3 + \infty$$

(ii) $\beta - 2 = 3 - 0 = 3$ No of loci = $3(\text{equat 6 No f})$
(iii) Real aris loci lie between 0 and -1 and
 $b/\omega - 3$ and $-\infty$
(iv) Thus there is one breakaway point, which,
can be found using angle-magnitude criteria
 $le' = KQ(s)H(s) = -1$
 $K = -\frac{1}{Q(s)H(s)} = -s(s+1)(s+3)$
 $K = -(s^3+4s^2+3s)$
 $\frac{dk}{ds} = -\frac{d}{ds}(s^5+4s^2+3s) = 0$

$$-(3s^{2}+8s+3)=0$$

$$3s^{2}+8s+3=0$$

from which S = -0.45 and -2.21, in which case only -0.45 lies on the real axis loci Thus Sb = -0.45. (5100 and -1)

(N) The centre of asymptote $\delta_{c} = \frac{\sum p - \sum 2}{p - 2} = \frac{0 - 1 - 3 - 0}{3} = -1.33$ $\therefore \quad \delta_{c} = -1.33$

(Vii) This means not locus will cross. jw-asa. The jw-crossing is famed by considering Q(s) in BCLT/F a.

$$CLT/F = \frac{K}{s(s+i)(s+3)+K}$$

$$= \frac{K}{s^{3}+4s^{2}+3s+K}$$

$$Q(s) = s^{3}+4s^{2}+3s+K$$

$$S=j\omega \quad Q(j\omega) = -j\omega^{3}-4\omega^{2}-j3\omega+K = 0$$

Equating energy part to zero.

$$-j\omega^{3} - j3\omega = 0.$$

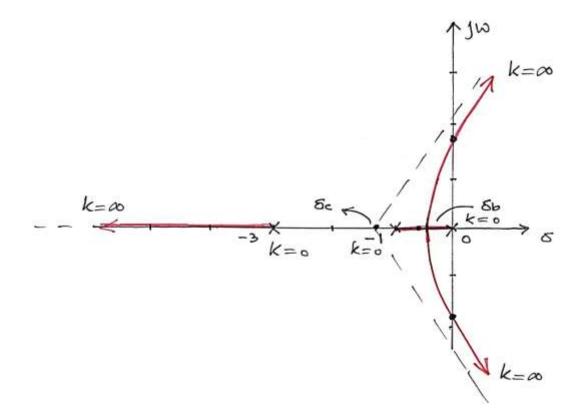
$$\omega^{3} = -3\omega$$

$$\omega^{2} = -3$$

$$\omega = \pm j1.732$$

$$\cdot j\omega - conserved is at + j1-732 and - j1-732$$
(VIII) Gove at jw - conserved is from equating real portion of $Q(j\omega) = 0.$

The root locus plotted is shown on the next page.



- 1. We can see that now the root locus is not a vertically straight line but it follows a curve and crosses the jw-axis. Thus adding a pure integrator will make the otherwise stable system to a system that may lead to instability if the operating point is not wisely selected.
- 2. The root locus, on adding a pure integrator in series with the plant will tend to shift the root locus more towards the right as can be noted from example 3 and 4 with their break-away points shifted to right in case of example 4. This means that the transient response in-terms of settling time will deteriorate. The settling time will increase in case when integrator is included and therefore the transients or disturbance will not die out quickly. This may be a bad news in the case of aero-plane auto pilot system.
- 3. However, shifting the root locus to the right improves steady-state performance since more gain is added into the system.
- 4. Adding an integrator in the forward path will also change the system-type. Comparing example 3 and 4 indicate that the system type changes from Type-0 (example 3) to Type-1 (example 4)

Conclusions:

1. Adding a pure gain derivative controller will improve the transient performance by shifting the original root locus to the left. In this cane the settling and rise time becomes less and the disturbances die out quickly (speedy transient response)

- 2. Adding a pure gain integrator controller will deteriorate the transient performance by shifting the root locus to the right. However, the steady-state performance is improved.
- 3. Including integrator changes the system type from lower to higher order.

However, practically pure gain derivative and integrator controller does not exist. Instead proportional derivative (PD) and proportional integrator (PI) controllers are used. These controllers are incorporated with an added gain element with their transfer functions as follows:

PD
$$G(s) = sK_D + K$$

And

PI
$$G(s) = \frac{K_I}{s} + K$$

A combination of the above two is the proportional integral derivative (PID) controllers with transfer function:

PID
$$G(s) = \frac{K_I}{s} + sK_D + K$$